The background of the slide is a photograph of the Apollo 16 Lunar Module (LM) in orbit above the lunar surface. The LM is a complex, multi-faceted structure with various instruments and antennas. The lunar surface below is dark and covered in numerous craters of various sizes. In the upper center, the blue and white horizon of the Earth is visible against the blackness of space.

The Kalman Filter

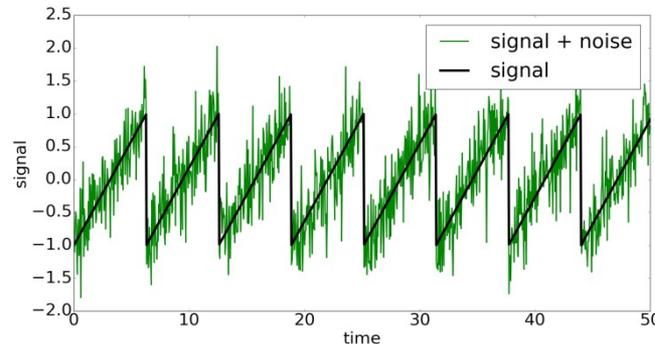
Eli Chertkov

Algorithms interest group talk

Source: nasa.gov

Some history of filtering

A central problem in signal processing is **filtering**, finding a signal in noise

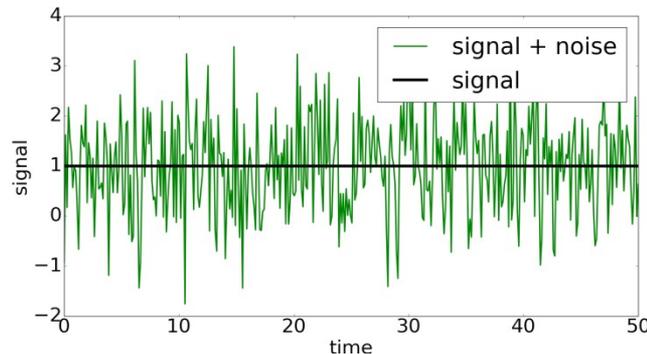


Source: Wikipedia



The **Wiener Filter** was proposed by Norbert Wiener and Andrey Kolomogorov independently in the 1940's. Wiener worked on applying the filter to the

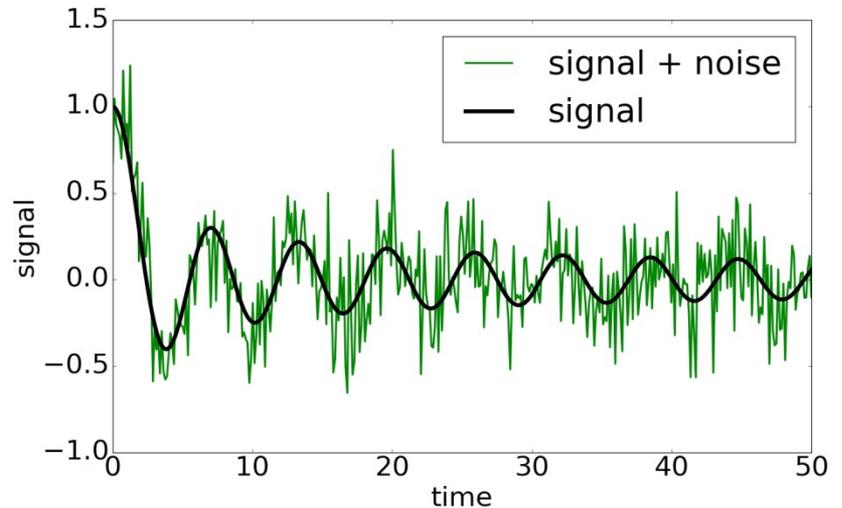
aircraft guns in



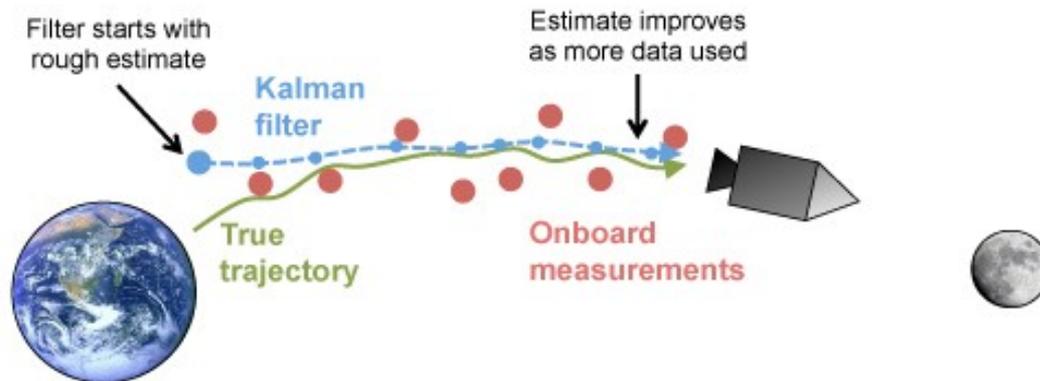
The Wiener Filter was designed for filtering continuous, stationary signals.

Some history of filtering

The Kalman Filter is an extension of the Wiener Filter to non-stationary (and usually discrete) signals.



Rudolf Kalman developed the **Kalman Filter** in the 1960's. The filter was used to help control spacecraft navigation systems. It was even used in the NASA Apollo missions to the moon!



Source:
<https://plus.maths.org/content/understanding->

The problem the Kalman Filter solves

The goal of the Kalman Filter is to estimate the time evolution of a system, whose **dynamics** obey a linear stochastic finite-difference equation

$$\mathbf{x}_t = \mathbf{F}_t \mathbf{x}_{t-1} + \mathbf{B}_t \mathbf{u}_t + \mathbf{w}_t$$

Current state of system

State transition matrix

Previous system state

Control to state conversion matrix

Control (driving forces)

Process noise

This estimate is based on **measurements**, which are linearly proportional to the system's state

$$\mathbf{z}_t = \mathbf{H}_t \mathbf{x}_t + \mathbf{v}_t$$

Measurement

Conversion matrix

System state

Measurement noise

Noise is assumed to be Gaussian distributed.

Sketch of derivation of Kalman Filter equations

The Kalman Filter algorithm involves iteratively computing a Gaussian estimate for the current state based on **1) previous estimates** and **2) measurements**.

1) Compute prediction from previous estimates

Estimated mean $\longrightarrow \hat{x}_{t|t-1} = F_t \hat{x}_{t-1|t-1} + B_t u_t$

Estimated covarianc $\longrightarrow \hat{P}_{t|t-1} = F_t \hat{P}_{t-1|t-1} F_t^T + Q_t$ \longleftarrow Covariance of process noise

e These equations come from inserting the equation of motion into definitions of mean and covariance.

2) Combine with current measurements

$$\hat{x}_{t|t} = \hat{x}_{t|t-1} + K_t (z_t - H_t \hat{x}_{t|t-1})$$
$$\hat{P}_{t|t} = \hat{P}_{t|t-1} - K_t H_t \hat{P}_{t|t-1}$$

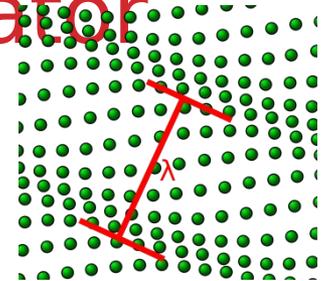
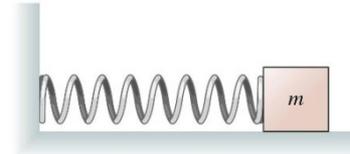
where $K_t = \hat{P}_{t|t-1} H_t^T (H_t \hat{P}_{t|t-1} H_t^T + R_t)^{-1}$ is the Kalman gain.

These equations come from adding process and measurement Gaussians together and completing the square.

Example: Simple Harmonic Oscillator

We all know the ubiquitous simple harmonic oscillator and its equation of motion:

$$\ddot{x}(t) = -\omega^2 x(t)$$



The SHO equation is a second-order linear ODE. We can write it as a system of first-order linear ODEs.

$$\begin{aligned} x_1 &\equiv x & \dot{x}_1 &= x_2 \\ x_2 &\equiv \dot{x} & \dot{x}_2 &= -\omega^2 x_1 \end{aligned}$$

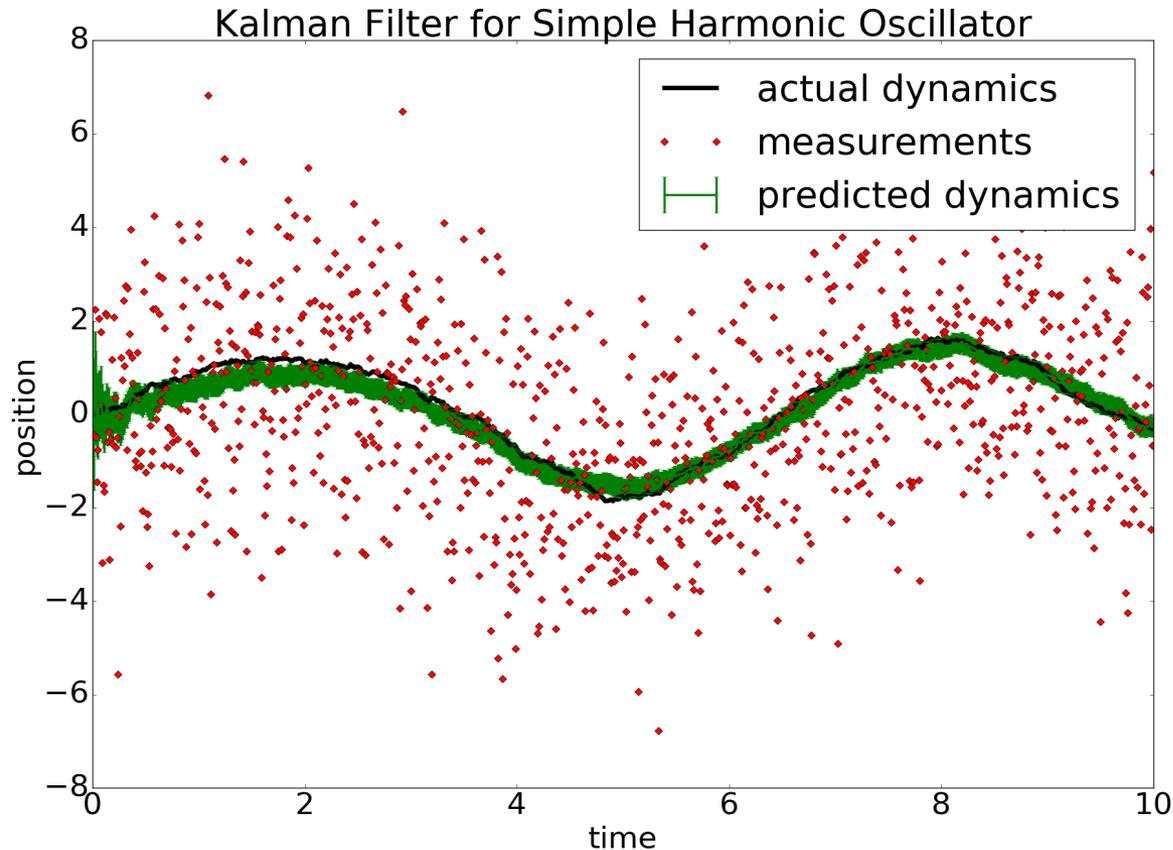
We can now discretize the ODEs with finite-differences to obtain

$$\begin{pmatrix} x_1 \\ x_2 \end{pmatrix}_t = \begin{pmatrix} 1 & \Delta t \\ -\omega_0^2 \Delta t & 1 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}_{t-1}$$

Now we can apply the Kalman Filter.

Example: Simple Harmonic Oscillator

Suppose that we have a SHO subject to random perturbations whose position we can measure with a noisy measuring device. How well can a Kalman Filter pick up the signal?



Example: Launching a satellite into orbit

Suppose that we are a telecommunications company that wants to launch a satellite into geosynchronous orbit.

We know the equations of motion of the satellite under the influence of gravity.



They are (in polar coordinates)

$$\ddot{r} + r\dot{\phi}^2 - \frac{u_r}{m} + \frac{GM}{r^2} = 0$$

$$2\dot{r}\dot{\phi} + r\ddot{\phi} - \frac{u_\phi}{m} = 0$$

Source:

<http://www.space.com/19593-amazing-rocket-launches-photos-2013.html>

These are non-linear! Does the Kalman Filter apply?

Yes! We can linearize it at each time step by computing the Jacobian. Then we can use finite-differences as before. This non-linear filter, which combines Jacobians with the Kalman Filter is called the Extended Kalman Filter.

Great, but what about the control parameters? How do we pick them?

Digression: PID control

Proportional-Integral-Derivative (PID) control is a simple, general purpose control scheme. The control signal (driving force) has three separate terms, each of which attempts to force the system state into a target state

$$u(t) = u_P(t) + u_I(t) + u_D(t)$$

Proportional control

$$u_P(t) = -K_P e(t)$$

Integrated control

$$u_I(t) = -K_I \int_0^t e(t') dt'$$

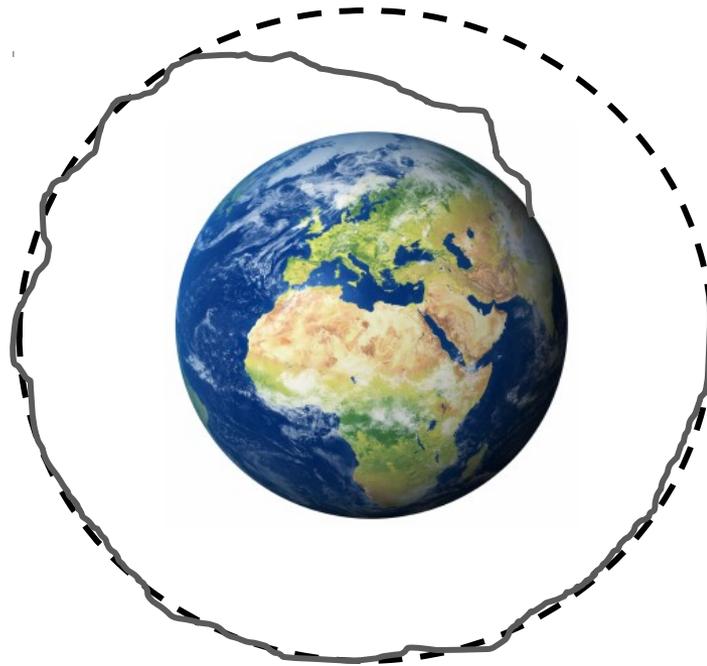
Derivative control

$$u_D(t) = -K_D \frac{de(t)}{dt}$$

where $e(t) = y(t) - x(t)$ is the error.

Example continued: Launching a satellite into orbit

Using PID control along with the Kalman Filter, we can guide a satellite from launch to a geosynchronous orbit, even with a noisy measuring apparatus.



Sketch of the result

Thank you!

You can find example scripts implementing the Kalman Filter in the ipython notebook posted on the group github.